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FAA Form 337 (12-88)

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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements. 8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) DI Installed Sky craft Wing Tip Landing Lights PN PCL 100 ILAIN STC SA 504 NE. 2) Installed Whelen wing Tip Strobes I/A/W STC SAGNE 1 cm A 650 PR NAV STROBE Light lea A 650 PG NAU STROBE Light 1 en HD-60 Installation Kit 1 en A-413-HDACF Power Supply 3 All Wining, Switches, and Cincit Protecting Devices are I/A/W AC 43.13-1 A, Chapter 11, Section 1, 2, 3, and 7. (4) Equipment List modified and weight & Balance bata Computed and Entered in Algoraft Log Bock. - END. 0-1 M-2 SPM C-1 C-2 A-1 M-1 0-2 A-3 0-3 0.4 A-4 APR _ 2 1997 A-5 0-5 A-6 0-6 COLUMBUS, 03 FSDO C5 A-10 A-9 A-7 C3 0.7 0-9 **C**4 A-8 0-8 Additional Sheets Are Attached

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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) Removed LRN500 mounting tray from left hand radio stack 75" AFT of Datum. Installed Garmin GPS-150 in left hand radio stack 77" AFT of Datum. Installed Garmin GA-56 GPS antenna on top fuselage 110" AFT of Datum. GPS 150 wired per Garmin GPS-150 Installatikon Manual P/N 190-00026-00 Rev. K. GPS-150 installed as a stand alone system. GPS-150 protected by a 3 amp circuit breaker labeled GPS Panel Placarded "GPS NOT APPROVED FOR IFR" Ground test performed I.A.W. A.C. 20-138. Garmin GPS-150 Pilots Guide P/N 190-00048-00 Rev. D placed in Aircraft. All work performed I.A.W. A.C. 43.13-2 Chapter 1, 2, 3, Section. 1A, STC SA0006WI. Electrical Advisory Circular A.C. 20-138 and reference load computed and does not exceed the capability of the Aircraft Electrical System. Weight and Balance computed and entered in the Airframe Logbook.

END_

Additional Sheets Are Attached

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